$(\textbf{KEY CONCEPT}) \ [\textbf{C-2-3}]$ Stagnation Point: p_0, M_0 Minimum Pressure Point: p, M When M = 1 at the minimum pressure point, this becomes "critical" point $C_p = C_{p,cr}$ and $M_\infty \to M_{cr}$ Freestream: p_∞, M_∞ Wing and airfoil sketch (source: J.D. Anderson "Fundamentals of Aerodynamics" 2016)

